# DESIGN AND CONSTRUCTION OF A TORSIONAL FATIGUE TESTING MACHINE OPERATED BY INERTIAL LOADS

# DISEÑO Y CONSTRUCCIÓN DE UNA MÁQUINA PARA PRUEBAS DE FATIGA TORSIONAL OPERADA BAJO CARGAS DE ORIGEN INERCIAL

# FABIO MARCELO PEÑA BUSTOS

M. Sc, Universidad Autónoma de Manizales, fabiope@autonoma.edu.co

# CÉSAR AUGUSTO ÁLVAREZ VARGAS

M. Sc, Universidad Autónoma de Manizales, dekinov@autonoma.edu.co

Received for review October 27 th, 2011, accepted December 3th, 2010, final version January, 4 th, 2011

**ABSTRACT:** In this paper the development of a torsional fatigue testing machine for metallic materials is presented. This machine is considered type I according to the Weibull classification because the fatigue failure is caused by a load of inertial origin. The machine operation is based on the alternating movement that the transmission of a washing machine produces. The inertial load is caused by a flywheel located at the top of the machine. Unlike many machines that generate fatigue-alternating stresses through relocation or the rotation of the specimen on loads that are consistent across time, the design of the machine clearly shows how inertial loads due to the acceleration of the mechanical elements generated loads of alternating character which eventually causes fatigue failures.

**KEYWORDS:** Fatigue, torsional fatigue testing machine, inertial loads

**RESUMEN:** En este artículo se presenta el desarrollo de una máquina para pruebas de fatiga torsional en materiales metálicos. La máquina se considera de tipo I según la clasificación de Weibull debido a que la falla es producida por una carga de origen inercial. El principio de operación se basa en el movimiento alternante que produce la transmisión de una máquina lavadora. La carga inercial es producida por un volante ubicado en la parte superior de la máquina. A diferencia de muchas máquinas de fatiga que generan la alternancia de los esfuerzos mediante la reubicación o giro de la probeta respecto a cargas que son uniformes en el tiempo, en el diseño realizado se evidencia cómo las cargas inerciales debidas a las aceleraciones de los elementos mecánicos generan cargas de carácter alternante que llegan a ocasionar fallas por fatiga.

PALABRAS CLAVE: Fatiga, máquina para pruebas de fatiga torsional, carga de origen inercial

#### **1. NOMENCLATURE**

A : Amplitude ratio

c: Effective radius of the testing specimen

 $l_{00}$ : Moment of inertia regarding the rotation axis

*I*: Polar moment

 $K_f$ : Dynamic stress concentration factor

*N<sub>k</sub>*: Cycles to failure

R: Stress ratio

[*r<sub>c</sub>*]: Correlation matrix

r<sub>i</sub>: Primitive radii

 $R_i$ : Euclidian norm of the position vector in the openconfiguration

 $R_i$ ': Euclidian norm of the position vector in the crossconfiguration  $R_i$ : Position vector at the opened configuration

 $R_i$ : Position vector at the crossed configuration

**T**: Torque

## **Greek symbols**

 $\alpha_i, \alpha_i'$ : Magnitude of the angular acceleration

 $\alpha_{i\alpha}$ : Alternating component of the magnitude of angular acceleration i

 $\alpha_{im}$ : Mean component of the magnitude of angular acceleration i

 $\theta_i$ : Director angle of the position vector in the openconfiguration

 $\theta_i'$ : Director angle of the position vector in the crossconfiguration

47

 $\mu$ : Erdman-Sandor configuration factor  $\varphi_i$ : Angle between two vectors  $\varphi_i'$ : Angle between two vectors when at least one of them refers to a position in the cross-configuration  $\sigma_{VM}$ : Von Mises effective stress  $\omega_i, \omega'_i$ : Angular speed of the mechanism links  $\omega_k$ : Angular speed of the AC motor

2. INTRODUCTION

Most of the available fatigue properties on materials come from the rotating-beam tests and from axial loading tests; however, little data are obtained from torsional tests and most of them are for materials that behave as ductile under static loading [1]. This fact is peculiar since mechanical components in many industrial applications are operated under conditions of variable torsion load [2,3]. Much torsion fatigue data are needed to determine the constants of materials used in multiaxial fatigue designs [4].

To obtain more accurate information about torsional fatigue properties in metallic materials, a torsional fatigue machine was built at the *Universidad Autónoma de Manizales*. This machine uses inertial loads produced by a flywheel to cause the failure. In this paper, a general description is presented about the machine operation as well as the mathematical formulation used to determine the fully reversed wave shape that rules the testing specimen movement. The paper also shows the general behavior of the failure specimens in the initial testing and its relationships to the data found in the literature.

# **3. THE MACHINE OPERATIONAL PRINCIPLE**

## 3.1. Torsional fatigue machines classification

According to Weibull [5], the machines used to perform torsional fatigue tests are classified depending on the type of load used to generate the failure. Each machine is identified according to the following nomenclature:

- Machine Type I: load caused by mechanical deflection and load caused by inertial forces
- Machine Type II: load caused by dead weights
- Machine Type III: load caused by centrifugal forces
- Machine Type IV: load caused by electromagnetic forces

• Machine Type V: load caused by pneumatic forces and load caused by hydraulic forces

Several kinds of machines have been built to characterize the mechanical behavior of materials that are subject to torsional fatigue. These machines include the one developed by Wöhler (Type I, 1871); Moore and Kommers (Type II, 1927); Bohuszewicz and Späth (Type IV, 1928); Lehr and Ruef (Type III, 1943); and the one developed by Quinlan (Type V, 1946) [5]. This list should also include the machine built by Berry (1956), through which one of the first studies on the torsional fatigue of aluminum alloys [6] came into existence; the machine built by Hussain (2000) who designed an economical machine for torsional fatigue using a servo-actuator [7]; and the machine built by Tonom, Barbieri and Barbieri (2009) that uses a resonance system to induce the failure [1].

The machine built in the *Universidad Autónoma de Manizales* is type I because it uses inertial loads to generate the specimen fracture; this failure is caused by a flywheel of pre-planned dimensions that generates inertial opposition to the alternating movement of the test specimen.

Fig. 1 shows a scheme of the specimen-flywheel system in pure rotation with respect to an axle. The system is fed by a variable function of the magnitude of the angular acceleration  $\alpha_5$  during the oscillation. From the kinetic analysis, the torque acting on the testing specimen at the instant under consideration is

 $T(t) = I_{00}\alpha_5 \qquad (1)$ 

where  $I_{00}$  is the moment of inertia of the rotating elements located from the testing specimen up to the flywheel.



Figure 1. Functional representation of the testing specimen–flywheel system

As will be shown in the kinematic analysis, the magnitude of the angular acceleration  $\alpha_5(t)$  is a function of the angular position of the crank and is also a function of the AC motor angular speed. Therefore, the shear stress wave generated over the testing specimen surface in a single cycle can be estimated from the drive mechanism kinematic analysis of the machine and can be controlled by the manipulation of the AC motor frequency and by the selected flywheel. This analysis assumes that the dynamic effects of rotation are independent of the specimen's torsional elasticity and independent of the rotational elements of the machine.

Such independence is achieved by raising the torsional stiffness of the rotational elements of the machine. The elevation of the torsional stiffness is also necessary in order to ensure that the elastic system specimenflywheel operates on a level which is lower than its fundamental torsional frequency.

Next, we will show how the alternating movement of the testing specimen is generated.

#### **3.2.** General description of the machine

In Fig. 2, a schematic representation of the machine designed can be seen. The machine consists of two assemblies: the *lower* ranging from the AC motor to the lower sliding clamp where the testing specimen is subjected, and the *superior* ranging from the upper clamp to the flywheel which generates the inertial load.



Figure 2. Description of the torsional fatigue testing machine designed at the *Universidad Autónoma de Manizales* 

The main elements of the machine, which are used to carry out its purpose, are:

- A three-phase asynchronous squirrel-cage electric motor. Power: 372.8 W; maximum angular speed: 1,750 r/min; frequency: 60 Hz; electrical potential difference: 220 V.
- Flexible belt-drive system. A35 V-belt. Transmission ratio: 1.136:1.
- Two-stage gear reducer. Transmission ratio: 8.58:1
- Self-alignment nuts and collets were used to hold the testing specimen. The nuts diameters were 7.938 mm (5/16"); 9.525 mm (3/8") and 12.7 mm (1/2").
- Two flywheels were designed and built to generate the inertial load: one in AISI 1020 steel (Ioo = 7.907 × 10<sup>-3</sup> kg · m<sup>2</sup>), used on a preferential basis to test steel specimens and one in 96.97% Al, 1.45% Si extruded aluminum (Ioo = 1.324 × 10<sup>-3</sup> kg · m<sup>2</sup>), used on a preferential basis to test aluminum alloy specimens.
- Variable-frequency drive (VFD) Altivar 31. Power: 746 W. The VFD can adjust the AC motor circular frequency between 600 r/min and 2,400 r/min and allows the generation of a ramp function for the speed at startup.
- Telemecanique XS608B1P inductive sensors. Diameter: 8 mm; 12/48 VDC. Two sensors are used: one to count the number of cycles and one to automatically turn off the machine when the testing specimen is broken.

# 4. KINEMATICAL DESCRIPTION OF THE MACHINE

The alternating rotational movement of the testing specimen is generated by a plane four-bar crank and rocker linkage; this mechanism was adapted from the transmission of a commercial washing machine.



**Figure 3.** a) Representation of the four-bar linkage that operates the machine, b) the four-bar linkage components. The clamp that supports the testing specimen is joined to a pinion (not shown) that is geared with output link 3 (the rocker) that produces the fully reversed movement.

Figure 3 shows the CAD mechanism as well as its symbolic description; this representation is needed for the kinematic analysis. The crank of the mechanism is the output gear of a two-stage speed reducer.

Figure 4 shows the mechanical drive system with the identification of its corresponding elements. The cyclical movement of the testing specimen is generated by a pinion connected to the rocker. The total transmission ratio from the AC motor to the testing specimen is 9.89:1; this value includes the transmission ratio of the belt drive system and the transmission ratio of the two-stage speed reducer.

In order to estimate the magnitude of the angular acceleration of the testing specimen, the equation for the analytical method of vector loop proposed by Norton [8] was used. Figure 5 illustrates the vector loops and the variables used for the development of the kinematic analysis, and it also shows the two possible configurations of the four-bar linkage: the crossed-configuration and the opened-configuration.



Figure 4. Driving mechanism of the machine

With the variables illustrated in Fig. 6, it is possible to establish the geometric relationships that describe the kinematics of the mechanism; so that the following equations describe the position analysis:

$$R_1 + R'_4 = R_2 + R_3'$$
 (2)  
 $R_5 + R_3' = R_4'$  (3)

Equations (2) and (3) are the vector loops that configure the four-bar mechanism position. Vectors  $\mathbf{R_1}$  and  $\mathbf{R_2}$ reference the positions of the main frame and the crank, respectively; vectors  $\mathbf{R_3}'$  and  $\mathbf{R_4}'$  reference the positions of the coupling link and the output link (rocker), respectively; and vector  $\mathbf{R_5}$  is an auxiliary vector that is used to differentiate between the linkage opened-configuration and the linkage crossedconfiguration. All vectors are referred to with regard to the Galilean framework "XY" as seen in Fig. 5. Using complex compact notation, the vectors used in the analysis can be expressed as

$$\boldsymbol{R}_i = R_i e^{j\theta_i} \qquad (4)$$

with

$$i = 1, 2, 3, 4, 5$$

or

$$\mathbf{R}_{i}' = R_{i}' e^{j\theta_{i}'} \tag{5}$$

with i' = 1, 2, 3, 4i' = 1, 2, 3, 4



Figure 5. The driving-mechanism position vectors

where the prime is used to distinguish between position vectors in the crossed-configuration and the opened-configuration (see the nomenclature),  $R_i$  represents the Euclidian norm of the position vector,  $\theta_i$  represents the angle of the position vector, and j is the imaginary unit.

From Eqs. (2) to (5), Euler's exponential formula, and the law of cosines, the following equations which determine the main angular magnitudes can be established:

$$\begin{split} \varphi_{1} &= \cos^{-1} \left( \frac{R_{4}^{2} + R_{5}^{2} - R_{3}^{2}}{2 R_{4} R_{5}} \right) \quad (6) \\ \varphi_{1}' &= \cos^{-1} \left( \frac{R_{4}'^{2} + R_{5}^{2} - R_{3}'^{2}}{2 R_{4}' R_{5}} \right) \quad (6.1) \\ \theta_{4}, \theta_{4}' &= \theta_{5} - \mu(\varphi_{1}, \varphi_{1}') \quad (7) \end{split}$$

where  $\mu$  is the Erdman-Sandor configuration parameter defined as

$$\mu = \begin{cases} -1, \text{ for the crossed} - \text{configuration} \\ 1, \text{ for the opened} - \text{configuration} \end{cases}$$
(8)

Considering Eqs. (7) and (8); vectors  $\mathbf{R_4}$ ,  $\mathbf{R'_4}$ , and  $\mathbf{R_5}$ ; and Euler's formula, the following equation is obtained:

$$\theta_{3}' = \tan^{-1} \left( \frac{R_{4}' \sin \theta_{4}' - R_{5} \sin \theta_{5}}{R_{4}' \cos \theta_{4}' - R_{5} \cos \theta_{5}} \right)$$
(9)

The input speed of the AC motor is transmitted to the crank of the four-bar linkage by the V-belt transmission and the two-stage speed reducer as shown in Fig. 2. Considering the angular speed of the crank  $\omega_2 \omega_2$  and the time derivatives of Eqs. (7) and (9), we obtain

$$\omega_{3}' = \frac{\omega_{2}R_{2}\sin(\theta_{4}' - \theta_{2})}{R_{3}'\sin(\theta_{3}' - \theta_{4}')}$$
(10)

$$\omega_{4}' = \frac{\omega_{2}R_{2}\sin(\theta_{2} - \theta_{3}')}{R_{4}'\sin(\theta_{4}' - \theta_{3}')}$$
(11)

Considering the time derivatives of Eqs. (10) and (11), the following relations for the magnitude of the angular accelerations are obtained:

$$\alpha_{3}' = \frac{CD - AF}{AE - BD}$$
(12)  
$$\alpha_{4}' = \frac{CE - BF}{AE - BD}$$
(12.1)

where

$$A = R_4' \sin \theta_4';$$

$$B = R_{3}' \sin \theta_{3}';$$

$$C = \alpha_{2}R_{2} \sin \theta_{2} + \omega_{2}^{2}R_{2} \cos \theta_{2} + \omega_{3}'^{2}R_{3}' \cos \theta_{3}' - \omega_{4}'^{2}R_{4}' \cos \theta_{4}';$$

$$D = R_{4}' \cos \theta_{4}' \qquad (13)$$

$$E = R_{3}' \cos \theta_{3}';$$

$$F = \alpha_2 R_2 \cos \theta_2 - \omega_2^2 R_2 \sin \theta_2 - \omega_3'^2 R_3' \sin \theta_3' + \omega_4'^2 R_4' \sin \theta_4'$$

The  $\alpha_2$  variable represents the magnitude of the angular acceleration of the input link driving mechanism. This is attenuated during starting by the variable-frequency drive speed controller that generates a ramp function from 0 to the desired angular speed in 30 s. From that time, the angular speed of the AC motor remains unchanged for the rest of the test. This fact is illustrated in Fig. 6.



Figure 6. Ramp input functions generated by the variablefrequency drive that controls the machine AC motor during the early 30 s

The  $\omega_k$  quantities shown in Fig. 6 correspond to the angular speeds of testing on the AC motor in which the first steel test specimens failed. These quantities were  $\omega_1 = 1,289$  r/min,  $\omega_2 = 1,237$  r/min,  $\omega_3 = 1,185$  r/min,  $\omega_4 = 1,151$  r/min, and  $\omega_5 = 1,091$  r/min. These speed values were estimated from the theoretical model developed from the stress-life approach and the failure theory of Hencky-von Mises [9,10,11], preventing the number of cycles of the first tests from being too high.

The value  $\omega_1$  was estimated to fail at 10<sup>3</sup> cycles.

Since the test specimen is connected to the pinion gear meshed with rocker output 3 (see Figs. 3 and 4), its angular acceleration can be determined as:

$$\alpha_5 = \left(\frac{r_2}{r_1}\right) \alpha_4' \qquad (14)$$

where  $r_2 = 50$  mm is the pinion primitive radius and  $r_1 = 17$  mm is the primitive radius of rocker.

Figure 7 shows the graph of the  $\alpha_5 \alpha_5$  magnitude according  $\theta_2 \theta_2$  crank angle for an input speed of 1,750 r/min.



Figure 7. Magnitude of the angular acceleration of the testing specimen

#### 5. CHARACTERIZATION OF WAVE STRESSES IMPOSED ON THE TEST SAMPLES

According to the Hencky-von Mises theory for a torsion system, the stress value that causes failure can be estimated as

$$\sigma_{VM} = K_f \sqrt{3} \ \frac{Tc}{J} \tag{15}$$

where c is the radius of the specimen and the other variables were defined in the nomenclature. From Eq. (1) and Eq. (15), we obtain that

$$\sigma_{VM} = K_f \sqrt{3} \frac{c}{J} I \alpha_5 \qquad (16)$$

Because  $K_f$ , c, J, and I values remain constant throughout the operation, it is concluded that the stress

wave is a scaled function of the wave shown in Fig. 7, since in the analysis of fatigue without corrosion, the material's response is unaffected by the waveform but it is affected by peak values of stress [12].

From Fig. 7 it is possible to obtain the maximum and minimum values of the magnitude of the angular acceleration in  $\theta_2|_{\alpha_{smax}} = 15^\circ$  and  $\theta_2|_{\alpha_{smin}} = 181^\circ$  which are independent of the AC motor angular velocity. Figure 8 shows the dependence of the peak and average values of  $\alpha_5$  per cycle, depending on the angular speed of the AC motor.

While the moment of inertia remains constant, Fig. 8 represents the torque's peak and the average applied values, depending on the flywheel used. For the machine designed, the maximum applied range of the torques were from 0.879 N·m to 10.38 N·m when using the steel flywheel, and between 0.212 N·m and 2.508 N·m when using the aluminum flywheel. The minimum torque increment that can be achieved with minimum frequency variation are 0.03 N·m for the first case and 0.01 N·m for the second case. Given that the magnitudes of the stresses are proportional to the magnitude of the angular accelerations, we have

$$\alpha_{5a} = \frac{\alpha_{5max} - \alpha_{5min}}{2} \quad (17)$$
$$\alpha_{5m} = \frac{\alpha_{5max} + \alpha_{5min}}{2} \quad (18)$$

where  $\alpha_{5a}$  and  $\alpha_{5m}$  were defined in the nomenclature.



Figure 8. Maximum magnitude, mean magnitude, and minimum magnitude of the angular acceleration of the testing specimen

From Eqs. (17), (18) and Fig. 7, it is possible to deduce that the values of stress ratio and amplitude ratio are, respectively:

$$R = \frac{\alpha_{5min}}{\alpha_{5max}} = -0.915 \quad (19)$$
$$A = \frac{\alpha_{5a}}{\alpha_{5m}} = 22.59 \quad (20)$$

Since  $R \approx -1$  and  $A \gg 1$ , it can be seen that the specimen is subject to a completely alternating stress wave [13]. The  $\alpha_{5a}$  regression depending on the angular speed of the AC motor can be expressed as:

$$\alpha_{5a} = 2.4 \times 10^{-4} \left(\omega^k\right)^2 - 3.5 \times 10^{-15} \omega_k \ (21)$$

with a correlation coefficient nearly equal to 1.

#### 6. EXPERIMENTAL TESTS

To test the machine, 24 AISI 1045 steel specimens were tested whose scheme is shown in Fig. 9. Its features are:

- Yield strength Sy = 1,229 MPa
- Ultimate strength Sut = 1,326 MPa
- Heat treatment: quenched and tempered to 41 HRC (Rockwell C hardness)
- Test diameter = 4.7 mm

 Table 1. Number of failure cycles obtained at the 5 groups of angular speeds

Group	ω <sub>k</sub> [r/min]	$N_k$	Log N <sub>k</sub>
1	1,289	1,190	3.075
	1,289	907	2.957
	1,289	678	2.831
	1,289	963	2.983
	1,289	954	2.979
2	1,142	2,496	3.397
	1,142	1,275	3.105
	1,142	8,299	3.919
	1,142	3,383	3.529
	1,142	1,747	3.242
3	1,022	755	2.877
	1,022	18,210	4.260
	1,022	4,217	3.625
	1,022	2,181	3.338
	1,022	3,388	3.529
4	960	2,986	3.475
	960	5,654	3.752
	960	5,222	3.717
	960	1,514	3.180
	960	2,789	3.445
5	861	9,827	3.992
	861	27,162	4.433
	861	7,841	3.894
	861	32,800	4.515

The specimens were subjected to 5 levels of angular speed (see Fig. 6). The number of specimens selected was based on ASTM E739-91 [14]. The results of these tests are illustrated in Table 1.



Figure 9. Geometric representation of the testing specimens used



Figure 10. Groups of angular speeds vs. number of cycles to failure



Figure 11. Regression line obtained from the tests data

Figure 10 shows the relationship between the AC motor angular speed  $\omega_k$  and the number of cycles to failure  $N_k$ , and Fig. 11 shows the same relationship in a logarithmic scale and the regression line of test data, whose equation is:

$$Log \omega_k = -0.09232 \ Log N_k + 3.3458$$
 (22)

The correlation matrix for the data obtained is:

$$[r_c] = \begin{bmatrix} 1 & -0.7388 \\ -0.7388 & 1 \end{bmatrix}$$

The correlation values show that there is adequate correspondence between the data tested and the regression line; although it is necessary to test a larger number of specimens to confirm this behavior. The standard deviation of the data on the regression line (vicinity of the data to the line) is 4.17% while the standard deviation of the data on the slope of the line is 0.37%. From the kinematic analysis and Eq. (16), we can obtain the Von Mises equivalent, alternating the stress values of each treatment. Figure 12 shows the relationship between applied stress and life in cycles of the material or S-N curve, following the methodology proposed by Wöhler [15].



1045 (41 HRC) steel

#### 7. CONCLUSIONS

A machine for dynamic torsional fatigue testing with inertial loads was designed and built, with the adaptability to fasten metal samples with diameters between 5 and 15 mm, and a maximum length of 100 mm. By controlling the input speed of the electric motor, it can apply a torsional wave pattern with fully alternating torque that can be adjusted in the 0.212

 $N \cdot m$  to 10.38  $N \cdot m$  range, depending on the flywheel selected. The estimation of alternating torque can be achieved through a kinematic analysis of the input mechanism according to the selected angular velocity of the AC motor.

Figure 13 shows a plot of normal probability of the logarithm of the number of cycles  $N_k N_k$ , since it is observed that the values clustered between quartiles 0.25 and 0.75 are close to a straight line, so it can be concluded that the data follow a normal or Gaussian probability distribution lognormal which is usual in steels [16], and this also shows that there was no point of atypical behavior that needs to be analyzed in a special way.



Figure 13.  $\log N_k$  probability normal plot

Preliminary testing on 24 specimens of AISI 1045 steel quenched and tempered to a final hardness of 41 HRC shows that the built machine allows for one to generate the S-N curve of material with coefficients of variation between 3.484 and 14.42% in Log  $N_k$   $N_k$  that are typical for fatigue testing [17], thus validating the dependence of torsional loading and therefore the equivalent of Von Mises stress as a function of the angular-velocity input mechanism.

Furthermore, regression of the data suggests a fatigue limit near the middle of the ultimate tensile strength (Fig. 13), which is typical behavior of steels [9].

## ACKNOWLEDGMENTS

The authors want to thank the engineer Raúl Alexander Mideros who manufactured the structural parts of the machine in his own workshop; the mechanical workshop coordinator in the UAM, Gilberto Andrés López López; and the mechanical engineering students Luis Fernando Fernández López, Camila Arango Villegas, Diana Lorena Cardona Montoya, Juan Camilo Martínez González, and Arturo Alexander Yépez Bracho who collaborated in the design, construction, adjustment, and the testing of the machine. We also give thanks for the valuable help of Mr. George Jiménez in the editing of this paper.

#### REFERENCES

[1] Tonon, J. M., Barbieri, R. and Barbieri, N., Investigating torsional fatigue with a novel resonant testing fixture. International Journal of Fatigue. Vol 31, pp. 1271-1277, 2009.

[2] Hussain, H. Torsion fatigue system for mechanical characterization of materials. Thesis for degree of Master of Science. Ohio University. Fritz J. and Dolores H. Russ College of Engineering and Technology, 33 P. Ohio, 2000.

[3] McClaflin, D. and Fatemi, A., Torsional deformation and fatigue of hardened steel including mean stress and stress gradient effects. International Journal of Fatigue. Vol 26, pp. 773-784, 2007.

[4] Kim, K. S., Chen, X., Han, C. and Lee, H. W., Estimation methods for fatigue properties of steels under axial and torsional loading. International Journal of Fatigue. Vol 24, pp. 783-793, 2002.

[5] Weibull, W., Fatigue testing and analysis of results. Belfast: Pergamon Press., pp. 36-38, 1961.

[6] Berry, J. W., Fatigue of aluminum as affected by temperature and intermittent periods of rest. Pasadena. Thesis for the degree of aeronautical engineer. California Institute of Technology. 1956.

[7] Hussain, H. Op. cit, pp.53-82.

[8] Norton, R. L., Design of machinery. An introduction to the synthesis and analysis of mechanisms and machines. Second Edition. New York: Mc Graw Hill., pp. 151-159, 1999.

[9] Norton, R. L., Machine design an Integrated Approach. Third Edition. New Jersey: Pearson Prentice Hall, pp. 315-342, 2006.

[10] Ibid, pp. 245-247.

[11] Cardona, D. L., Development of S-N torsion curve under inertial loads in AISI 1045 steel. Manizales, Universidad Autónoma de Manizales. 2008.

[12] Norton, R. L., Machine design an Integrated Approach. Op. cit, 309 P.

[13] Ibid., pp 305-306.

[14] AMERICAN SOCIETY FOR TESTING AND MATERIALS. Standard Practice for statistical analysis of

linear or linearized Stress-Life (S-N) and Strain-Life ( $\epsilon$ -N) fatigue data. West Conshohocken: ASTM, 2004. 7 p. (E 739-91).

[15] Bathias, C. and Paris, C., Gigacycle Fatigue in mechanical practice. New York: Marcel Dekker. Cap 4, 2005.

[16] Dowling, N., Mechanical behavior of materials. Second Edition. New Jersey: Prentice Hall. 367 P., 1999.

[17] Ibid., p. 803.