Analytical synthesis for four-bar mechanisms used in a pseudo-equatorial solar tracker

Síntesis analítica de los mecanismos de cuatro barras implementados en un seguidor solar pseudo – ecuatorial

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ABSTRACT

Photovoltaic energy production systems generate electricity without emitting pollutants into the atmosphere and do so from a free, unlimited resource. The highest level of energy conversion from the photovoltaic panels can be obtained by placing them perpendicular to the sun's rays falling on their surface; this is done by installing solar tracking systems. This work proposes the use of two fourbar mechanisms as the driving force for a solar tracker; we propose the use of analytical synthesis for such mechanisms. This procedure is aimed at optimising the transmission angle, increasing mechanical advantage and decreasing driving torque. A mathematical model was used to prove synthesis results and a prototype of the solar tracker was built.

Keywords: Transmission angle, four-bar mechanisms, solar tracker, synthesis.

RESUMEN

Los sistemas de producción de energía fotovoltaica generan electricidad sin la emisión de contaminantes a la atmósfera, a partir de un recurso libre e ilimitado. Con el fin de obtener el mayor nivel de conversión de energía de los paneles fotovoltaicos, éstos deben colocarse de manera perpendicular a los rayos del sol; esto se logra mediante la instalación de sistemas de seguimiento solar. En éste trabajo, se propone el uso de dos mecanismos de cuatro barras como impulsores de un seguidor solar, y el uso de un procedimiento analítico de síntesis para tales mecanismos. Dicho procedimiento tiene por objetivo optimizar el ángulo de transmisión, incrementar la ventaja mecánica y disminuir el torque impulsor. Se obtuvo, un modelo matemático para probar los resultados de dicha síntesis y se construyó un prototipo del seguidor solar.

Palabras clave: ángulo de transmisión, mecanismos de cuatro barras, seguidor solar y síntesis óptima.

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Introduction

There is a need to generate electricity from renewable resources nowadays; photovoltaic energy production systems use the photoelectric effect to transform solar energy into electricity, using photovoltaic cells placed on panels. Such energy conversion takes place without emitting pollutants into the atmosphere and also represents a free, unlimited resource (Tatu, 2010). However, photovoltaic systems do not compete with standard energy production systems yet, mainly due to three problems (Ochieng, 2010):

• Low efficiency, i.e. low energy production density per area installed;

• The relatively high cost of the materials involved in their construction; and

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• Variability during production, mainly related to changes in the weather.

One way of increasing energy conversion by 30% to 50% in photovoltaic energy production systems is the installation of solar tracking systems. This is because they must be placed perpendicular to the sun's rays falling on their surface to obtain the highest level of energy conversion from the photovoltaic panels; this is achieved through the use of solar trackers (Comsit, 2007). However, many photovoltaic energy production systems do not have tracking systems, mainly because of their cost and complexity (Duarte, 2010; Comsit, 2007). There is thus a need for developing efficient, low-cost, solar tracking systems.

It has been demonstrated that the best way to follow the path of the sun throughout the year is by using planar mechanisms, with their rotation axes arranged perpendicularly. This also simplifies

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the kinematic and dynamic analysis of the problem and has advantages regarding a control scheme and its implementation. As for solar trackers, research has been aimed at synthesising fourbar linkages with linear actuators (Visa, 2011), geared linkages, or developing new methods for evaluating their efficiency (Popa, 2008). Although in several referenced papers, dimensional synthesis for four-bar linkage (Comsit, 2007) and geared linkage (Creanga, 2010) has been carried out, analytic dimensional synthesis has not been used to improve transmission angle and decrease the motor's driving torque during design and construction phases. Specific methods for calculating the driving torque in such linkages (whose ultimate goal should be their construction and assembly) have not been presented in these sources. Our main objectives were thus to use analytical synthesis for two four-bar mechanisms and implement them in a pseudo-equatorial solar tracker. Our second aim was to obtain a mathematical model which would be used for proving that by optimising transmission angle, the peak torque required by these mechanisms' driving motors would become decreased, thus lowering the energy consumption of the whole tracking system and making it more efficient.



Figure 1: Pseudo-equatorial solar tracker: isometric view



Figure 2: Detail of diurnal axis driving mechanism

Description of the tracking system

The solar tracker had 10 photovoltaic panels (ERDM Solar 130TP6); they had a maximum power of 130 watts each. The pseudo-equatorial solar tracker had two axes of movement (see Figure 1): a diurnal axis (moving a 255 kg mass) and an elevation axis (moving an estimated 420 kg mass). It was installed in Tejalpa, Jiutepec, Morelos, Mexico. Pseudo-equatorial tracking systems use a local-observer angular system (with elevation and diurnal angles), while equatorial tracking systems use a global angular system.

The basic difference between equatorial and pseudo-equatorial trackers is the order of rotation of their axes (Burduhos et al., 2008). Two four-bar mechanisms were designed, built, assembled and coupled in series to drive the axes of movement of the pseudo-equatorial solar tracker, and hence the photovoltaic panels. The motors worked in a stepwise manner (Figures I and 2).

Synthesis of four-bar mechanisms

We proposed an analytical synthesis procedure for the four-bar mechanisms, known as Brodell and Soni procedure (Shigley, 1998). This procedure was aimed at optimising the mechanisms' transmission angle (angle γ , between coupler and rocker, which is a measure of the quality of transmission force and speed at the joint, complement angle being called the pressure angle). Several authors have suggested keeping this angle's values greater than 30° to ensure the highest mechanical advantage and reduce driving torque (Shigley, 1998; Norton, 1999). The procedure is described as reference in the following lines.

Figure 3 shows the geometric relationships for the four–bar mechanism: they meet Grashof's law, have a crank–rocker configuration and Q=1 speed ratio, i.e. when the crank goes from point A₁ to point A₂, 180° angular displacement occurs and when the rocker goes from point B₁ to point B₂ this produces total angular displacement ϕ (Figure 3a). Using Figures 3a) and 3b), and the cosine law, equations (1) – (4) were formed. These equations were in terms of γ_{min} and ϕ , and formed a system of linear equations, having unknowns r_1 , r_2 , r_3 and r_4 . Brodell and Soni proposed equation (5) as a condition to be met:

$$\cos(\theta_{4\min} + \phi) = (r_1^2 + r_4^2 - (r_2 + r_3)^2)/2r_1r_4 \qquad (1)$$

$$\cos\theta_{4,\min} = \left(r_1^2 + r_4^2 - (r_2 - r_2)^2\right)/2r_1r_4 \tag{2}$$

$$\cos \gamma = -(r^2 + r^2 - (r - r)^2)/2rr$$
 (3)

$$\begin{pmatrix} 2 & 2 & (&)^2 \end{pmatrix}$$

$$\cos \kappa = (r_3^2 + r_4^2 - (r_1 + r_2)^2)/2r_3r_4$$
(4)

$$\gamma_{\min} = 180^\circ - \kappa \tag{5}$$



Figure 3: a) Oscillation angle ϕ on link 4, b) minimum transmission angle and maximum angle formed at the intersection of coupler and rocker

$$[r_3 / r_1] = ((1 - \cos \phi) / (2 \cos^2 \gamma_{\min}))^{1/2}$$
(6)

$$[r_4 / r_1] = \left((1 - (r_3 / r_1)^2) / (1 - (r_3 / r_1)^2 \cos^2 \gamma_{\min}) \right)^{1/2}$$
(7)

$$[r_2 / r_1] = ((r_3 / r_1)^2 + (r_4 / r_1)^2 - 1)^{1/2}$$
(8)

where:

 ϕ rocker oscillation angle

- γ_{\min} minimum allowed transmission angle
- *κ* maximum angle formed at the intersection of coupler and rocker
- r_1 length [m] of link I, fixed
- r₂ length [m] of link 2, crank
- r_3 length [m] of link 3, coupler
- r_{A} length [m] of link 4, rocker

Values for the oscillation angles of link 4 (rocker) on both diurnal and elevation axes were proposed. These values can be seen in Figure 4. They are marked by Greek letter ϕ , because this letter denotes the angular displacement of the rocker in both mechanisms.



Figure 4: Proposed angles for link 4 (rocker), on the pseudo–equatorial solar tracker: a) diurnal axis, b) elevation axis

Oscillation angle, diurnal axis: $-50^\circ \le \phi \le 50^\circ$

Oscillation angle, elevation axis: $-20^{\circ} \le \phi \le 40^{\circ}$

Equations (6) - (8) were evaluated using Matlab R2010, letting ϕ be a fixed value, and varying ratios $[r_3/r_1]$, $[r_4/r_1]$, $[r_2/r_1]$ until γ became optimised, i.e. until transmission angle γ values were always greater than 30° and maximum values achieved for a full revolution of the crank, on both axes. These values were $39^\circ \leq \gamma \leq 90^\circ$ for the diurnal axis and $59^\circ \leq \gamma \leq 90^\circ$ for the elevation axis, proving the usefulness of this procedure as a means of synthesising four-bar mechanisms to be used in solar trackers.

Dynamics: driving torque

Although a kinetic-static approach would be enough for calculating shape, size and choosing materials for the prototype's structural members at low speed and acceleration, the dynamics of both mechanisms were calculated to evaluate the driving torque required by the motors.

A dynamic model of the four-bar mechanisms was obtained using Lagrange's formulation (Wu, 2002) (detailed in appendix A). Driving torque could thus be written as in equation (9):

$$T_{2} = 2(J_{1} + J_{2}S_{1}^{2} + J_{3}S_{2}^{2} + P_{1}C_{1}S_{1})\ddot{q} + [2J_{2}S_{1}(\partial S_{1} / \partial q + S_{1}(\partial S_{1} / \partial \theta_{3}) + S_{2}(\partial S_{1} / \partial \theta_{4})) +$$

$$+2J_{3}S_{2}(\partial S_{2}/\partial q + S_{1}(\partial S_{2}/\partial \theta_{3}) + S_{2}(\partial S_{2}/\partial \theta_{4})) +$$

$$+P_{1}(C_{1}(\partial S_{1}/\partial q + S_{1}(\partial S_{1}/\partial \theta_{3}) + S_{2}(\partial S_{1}/\partial \theta_{4})) +$$

$$+S_{1}(\partial C_{1}/\partial q + S_{1}(\partial C_{1}/\partial \theta_{3})))]\dot{q}^{2} +$$

$$-\partial G/\partial q - S_{1}(\partial G/\partial \theta_{3}) - S_{2}(\partial G/\partial \theta_{4}) - T_{4}(\dot{\theta}_{4}/\dot{q}) \qquad (9)$$

As a means of showing how driving torque decreased as transmission angle was optimised, equation (9) was evaluated for various values of r1, r2, r3 and r4 (resulting in various values for transmission angle, γ_{min} , as explained in the previous section), operating at constant speed. The required dynamic parameters were obtained after preliminary modelling of the mechanisms using Autodesk Inventor 2010.







Figure 6: Comparing driving torque for various transmission angle values for the elevation mechanism of the solar tracker



Figure 7: Transmission angle values for a full revolution of the crank

Figure 5 shows how the peak value of the driving torque required for the diurnal mechanism decreased as transmission angle value improved, until reaching γ_{min} =39°.

However, some dynamic parameters regarded as constant values (such as mass, inertia moment, and location of the centre of gravity of the links) underwent some variation as the synthesis results also varied; further studies will help determine whether the effect of such variation accounted for the relationship clearly seen in Figure 5 (diurnal mechanism), but not clearly enough in Figure 6 for the elevation mechanism. Figure 7 shows the transmission angle values for a full revolution of the crank in both mechanisms (diurnal and elevation).

Mechanical design: building a prototype

The wind load the solar tracker had to withstand was calculated, followed by the reaction forces on the joints and stress on links. Structural member shape, size and materials were assigned and a prototype was built.

Wind load

Regarding the loads a solar tracker must stand (weight of its own structure, weight of the photovoltaic panels, inertia load, wind load), critical load is due to wind. This is mainly because of the trackers' construction characteristics, i.e. their surface, height of installation, orientation regards wind direction (Butuc, 2010). Equations (10) - (12) were used to determine wind load (Messenger, 2004; Velicu, 2010).

$$F_{N} = \left(\rho A_{P} c_{f} v_{W}^{2}\right) / 2$$
 (10)

$$c_f = \left(c_{p1} + c_{p2}\right)/2 \tag{11}$$

$$T_4 = F_N d_0 \tag{12}$$

where:

F_N	normal force of wind on the surface of photovoltaic pan-
	els [N]
ρ	air density [kg/m³]

- A_p area of photovoltaic panels [m²]
- v_W wind speed [m/s]

/

- c_f force coefficient [unitless]
- C_{n1} pressure coefficient I [unitless]
- c_{p2} pressure coefficient 2 [unitless]
- distance from photovoltaic panel axis to the point of application of normal force, as a result of pressure distribution on the panel [m]
- T_4 external torque on link 4, due to wind load [N-m]

Figure 8 shows the point of normal force being applied to the photovoltaic panels.

The effect of wind load on the photovoltaic panels was calculated, and it was assumed that such a force would make link 4 rotate on its own axis, thus requiring greater motor driving torque to compensate for such external disturbance. It was proposed that the solar tracker must be able to move to its defence position, namely, having photovoltaic panels orientated horizontally, at wind speeds of up to 40 km/h (a wind speed of 6 on Beaufort scale, according to (http://www.spc.noaa.gov)). This criterion was used for calculating driving torque (Eq. 9).



Figure 8: Applying wind load to photovoltaic panel surface

It was statically assumed that the solar tracker, in one of its limit positions, must be able to withstand wind speeds of up to 80 km/h (a wind speed of 9 on Beaufort scale). This criterion was used for calculating stresses and deflections on links and selecting materials for the model.

Reaction forces on the joints: static case

The approach used for calculating stresses on the links of the mechanism was as follows. The reaction forces on the links and joints were found by considering each mechanism in one of its limit positions (statically), as they had to stand wind speeds of up to 80 km/h (winds capable of causing slight structural damage, according to (http://www.spc.noaa.gov)).

Figures 9 and 10 show the mechanisms in their limit positions as they were used for calculating reaction forces on their joints.



Figure 9: Schematic view of limit position for static analysis of diurnal mechanism



Figure 10: Schematic view of limit position for static analysis of elevation mechanism

Stresses and deflections on links

Stresses and deflections on the links were calculated, as a final step in the design stage. Figure 11 shows some of the results of these calculations, as verified using Autodesk Inventor 2010.



Figure 11: FEM model for coupler, diurnal mechanism

Experimental setup

Figure 12 shows the diurnal mechanism assembly, coupled to a worm–gear mechanism (also used as a break for the whole system) and to the actuating motor. As a result of this experiment, the angles theoretically predicted by the analytical synthesis (Figure 7) were proved. Variations ranging from 1% to 3% of theoretical values were observed, which could have been attributed to the manufacturing process.



Figure 12: Diurnal mechanism, whole assembly

Conclusions

According to Figure 5, the mathematical model for calculating the driving torque (given by equation (9)), proved that analytical synthesis for optimising the transmission angle of four-bar mechanisms, known as Brodell and Soni's procedure, could be successfully used in the synthesis of a pseudo-equatorial solar tracker's driving mechanisms.

Figure 5 shows a 40 N-m reduction for every 2° regarding peak value for the driving torque required for the diurnal mechanism; transmission angle γ_{min} value became improved.

Since driving torque became reduced, the proposed design led to a reduction in the energy consumption of the tracking system itself, thereby making the pseudo-equatorial solar tracker more efficient.

The procedure itself can be used for the synthesis and design of four-bar mechanisms to be used in any sized, very heavy, pseudo-

equatorial solar trackers and for various kinds of facilities.

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Appendix A

The dynamic model of the four-bar mechanisms was obtained using Lagrange's formulation (Figure 13).



Figure 13: Nomenclature for the four–bar mechanisms, using Lagrange's formulation

- r_i length of the link
- q generalised coordinate for the mechanism
- θ_i angles defining link position
- m_i mass of the links r_i
- I_i inertia moment of links r_i
- r_{CGi} distance from centre of gravity to pivot

Equations for position analysis were given by (13) - (17)

$$\theta_4(q) = 2 \arctan\left(\left(-k_1 \pm \sqrt{k_1^2 + k_2^2 - k_3^2}\right)/(k_3 - k_2)\right) \quad (13)$$

$$\theta_3(q,\theta_4) = \arctan((r_1 sen \theta_1 - r_2 sen q + r_4 sen \theta_4)) \cdot (14)$$

$$\cdot 1/(r_1 \cos \theta_1 - r_2 \cos q + r_4 \cos \theta_4))$$

$$k_1(q) = 2r_4(r_1 sen\theta_1 - r_2 senq)$$
⁽¹⁵⁾

$$k_2(q) = 2r_4(r_1 \cos \theta_1 - r_2 \cos q)$$
 (16)

$$k_{3}(q) = r_{1}^{2} + r_{2}^{2} - r_{3}^{2} + r_{4}^{2} - 2r_{1}r_{2}$$

$$(sen\theta_{1}senq + \cos\theta_{1}\cos q)$$
(17)

Equations for speed analysis are given by (18) - (20)

$$\begin{bmatrix} \dot{\theta}_3 \\ \dot{\theta}_4 \end{bmatrix} = \begin{bmatrix} S_1(q,\theta_3,\theta_4) \\ S_2(q,\theta_3,\theta_4) \end{bmatrix} \dot{q}$$
(18)

$$S_{1}(q, \theta_{3}, \theta_{4}) = \partial \theta_{3} / \partial q = r_{2} sen(\theta_{4} - q) / r_{3} sen(\theta_{3} - \theta_{4})$$
(19)

$$S_{2}(q, \theta_{3}, \theta_{4}) = \partial \theta_{4} / \partial q = r_{2} sen(\theta_{3} - q) / r_{4} sen(\theta_{3} - \theta_{4})$$
(20)

The total kinetic energy, potential energy and Lagrangian formulation of the system are given by equations (21) - (29).

$$T = (m_2 \|v_{CG2}\|^2 + I_2 \dot{q}^2)/2 + (m_3 \|v_{CG3}\|^2 + I_3 \dot{\theta}_3)/2 + (m_4 \|v_{CG4}\|^2 + I_4 \dot{\theta}_4)/2$$
(21)

$$V = m_2 g y_{CG2} + m_3 g y_{CG3} + m_4 g y_{CG4}$$
(22)

$$L(q,\theta_{3},\theta_{4},\dot{q},\dot{\theta}_{3},\dot{\theta}_{4}) = J_{1}\dot{q}^{2} + J_{2}\dot{\theta}_{3}^{2} + J_{3}\dot{\theta}_{4}^{2} + P_{1}C_{1}(q,\theta_{3})\dot{q}\dot{\theta}_{2} + G(q,\theta_{3},\theta_{4})$$
(23)

$$J_1 = \left(m_2 r_{CG2}^2 + I_2 + m_3 r_2^2\right)/2$$
(24)

$$J_2 = \left(m_3 r_{CG3}^2 + I_3\right)/2 \tag{25}$$

$$J_{3} = \left(m_{4}r_{CG4}^{2} + I_{4}\right)/2$$
(26)

$$P_1 = m_3 r_2 r_{CG3}$$
(27)

$$C_1(q,\theta_3) = \cos(q-\theta_3)$$
(28)

$$G(q, \theta_3, \theta_4) = (-m_2 g r_{CG2} - m_3 g r_2) \sin q + -m_3 g r_{CG3} \sin \theta_3 - m_4 g (r_1 \sin \theta_1 + r_{CG4} \sin(\theta_4 + \theta_{CG4}))$$
(29)

After finding the Lagrangian formulation, the equation of movement for the system (equation (9)) was found using equation (30):

$$V = m_2 g y_{CG2} + m_3 g y_{CG3} + m_4 g y_{CG4}$$
(30)